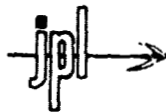


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FUSELAGE VENTILATION UNDER WIND CONDITIONS

**FIRE MODELING AND SCALING METHODS
510-56-05**



Jay Wm. Stuart



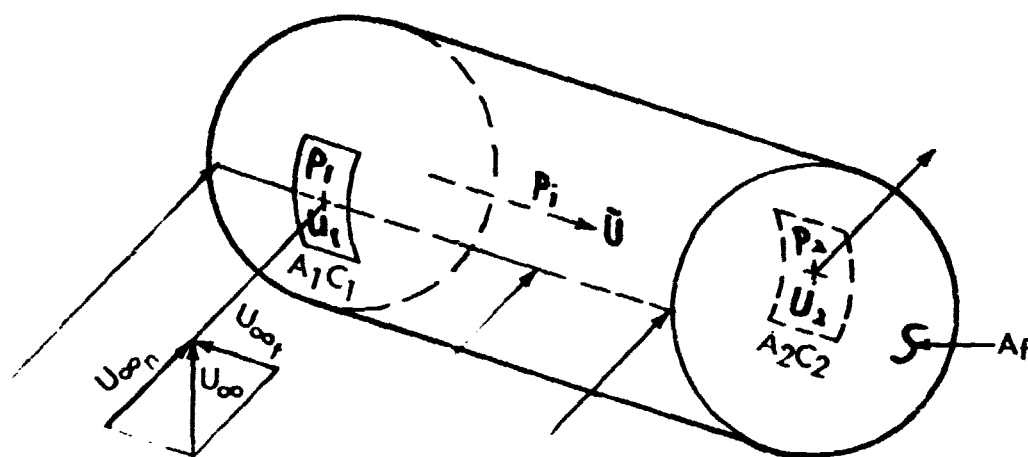
OBJECTIVES

- DETERMINE REALISTIC FUSELAGE VENTILATION RATES FOR
POST-CRASH FIRES AND FULL-SCALE FIRE TESTS

- FIND EFFECTS ON WIND-ABOUT-FUSELAGE VENTILATION RATE OF
VARIOUS PARAMETERS
 - FUSELAGE SIZE & SHAPE
 - FUSELAGE ORIENTATION & PROXIMITY TO GROUND
 - FUSELAGE-OPENINGS SIZE & LOCATION
 - WIND SPEED & DIRECTION



FLUID MECHANICS OF FUSELAGE VENTILATION



FROM MASS CONTINUITY AND ASSUMING $d\rho = 0$

$$\text{SOLVE } U_1 A_1 = U_2 A_2 \quad \text{OR} \quad A_1 C_1 \sqrt{2(p_1 - p_i)/\rho} = A_2 C_2 \sqrt{2(p_i - p_2)/\rho}$$

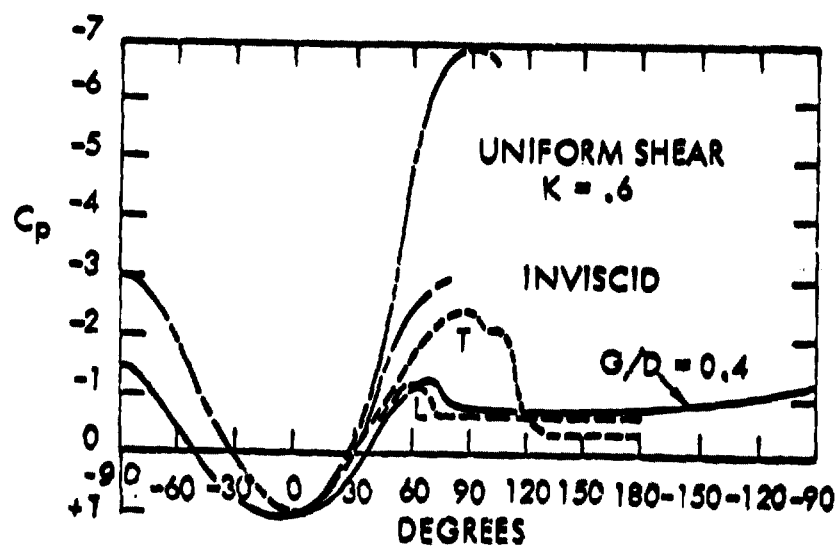
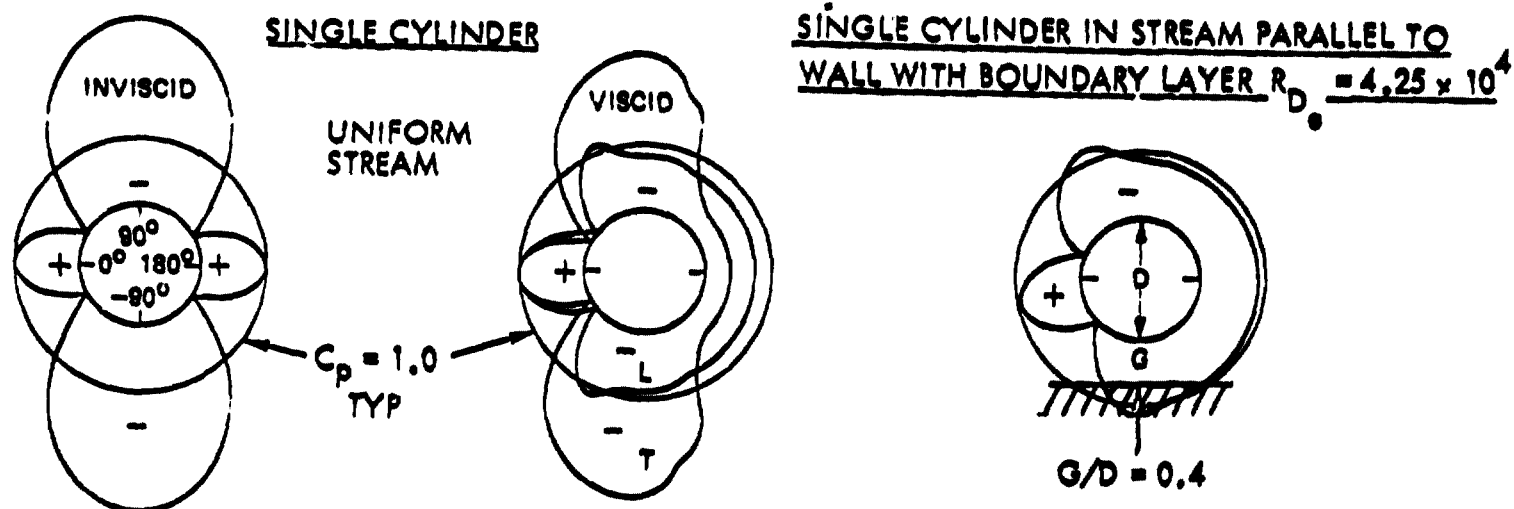
$$\text{LETTING } C_p = p/q, \quad q = \frac{\rho}{2} U_{\infty n}^2$$

$$\text{VOLUMETRIC RATE} \quad Q = C_1 A_1 U_{\infty n} \sqrt{C_{p1} - \left[C_{p1} + C_{p2} \left(\frac{A_2 C_2}{A_1 C_1} \right)^2 \right] / \left[\left(\frac{A_2 C_2}{A_1 C_1} \right)^2 + 1 \right]}$$

$$\text{INTERIOR VENTILATION SPEED} \quad \bar{U} = Q/A_f$$



PRESSURE DISTRIBUTIONS FOR FLOWS AROUND INFINITE CIRCULAR CYLINDERS





REFERENCES FOR PRESSURE DISTRIBUTIONS AROUND CIRCULAR CYLINDERS

1. FORREST E. GOWEN AND EDWARD W. PERKINS, "DRAG OF CIRCULAR CYLINDERS FOR A WIDE RANGE OF REYNOLDS NUMBERS AND MACH NUMBERS", NACA TN-2960, JUNE 1953
2. MELVIN H. SNYDER JR., "TESTING OF CYLINDERS IN SHEARED FLOW", J. AIRCRAFT, VOL. 8, AUGUST 1971
3. P. W. BEARMAN AND A. J. WADCOCK, "THE INTERACTION BETWEEN A PAIR OF CIRCULAR CYLINDERS NORMAL TO A STREAM", J. FLUID MECH. (1973), VOL. 61, PART 3, PP. 499-511
4. P. W. BEARMAN AND M. M. ZDRAVKOVICH, "FLOW AROUND A CIRCULAR CYLINDER NEAR A PLANE BOUNDARY", J. FLUID MECH. (1978), VOL. 89, PART 1, PP. 33-47



VENTILATION PERFORMANCE COMPARISON FIXED OPENINGS

$$A_1 = 2 \text{ m}^2 \text{ \& } C_1 = 0.6$$

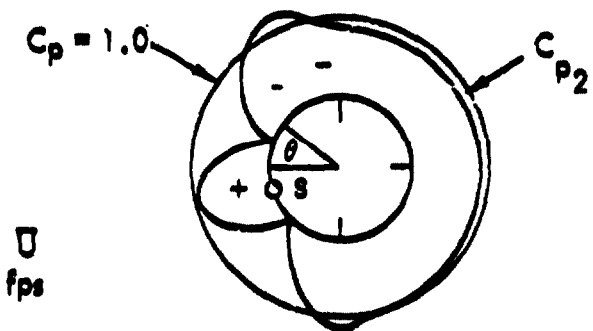
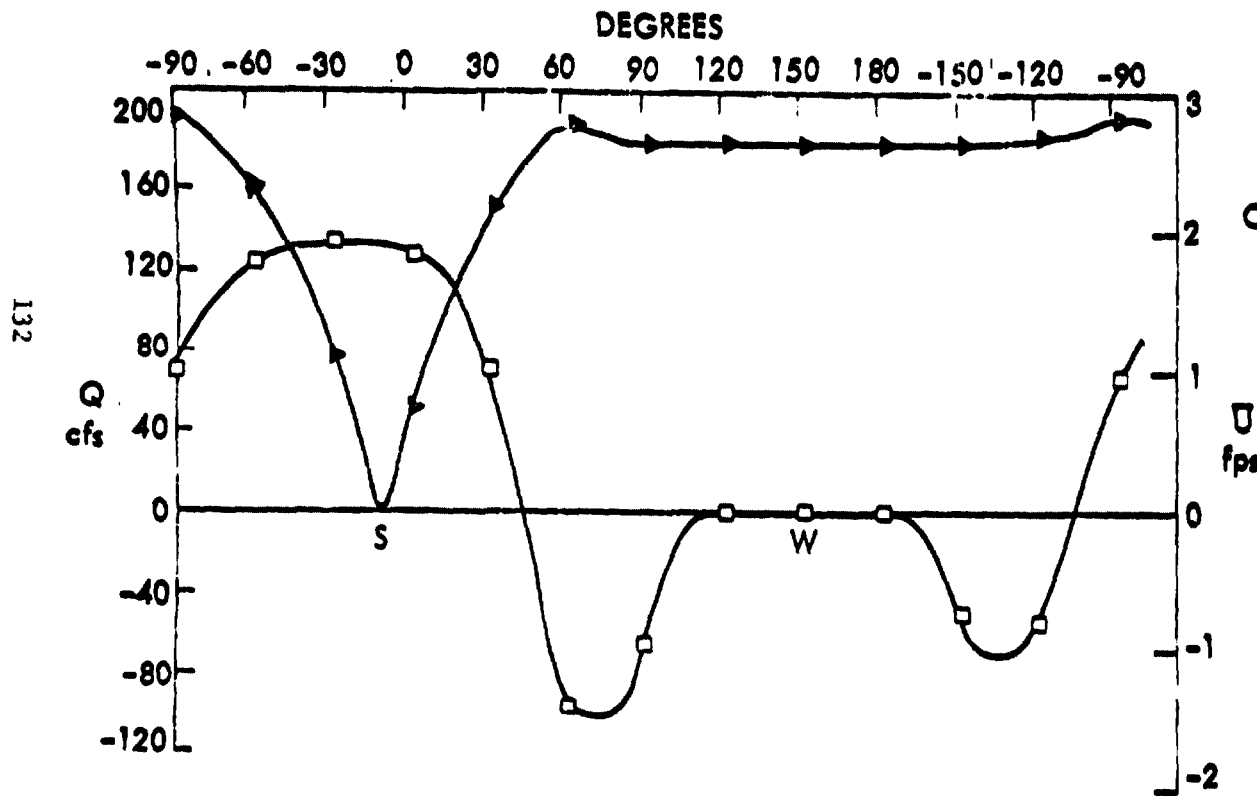
$$A_2/A_1 = 1.0 = C_2/C_1$$

$$U_{\infty} = 10 \text{ mph}$$

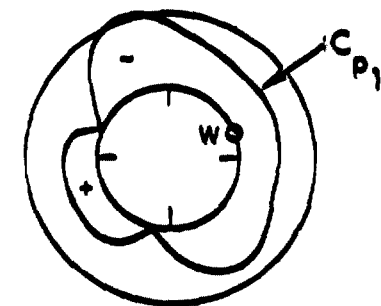
$$R_{De} = 4.8E4$$

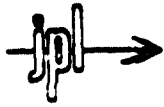
$$R_D = 1.23E6$$

$$\frac{G/D}{\text{FIXED OPENING}} \frac{C_p(\theta)}{\text{STAGNATION, } C_{p1} - C_{p2}(\theta)}$$

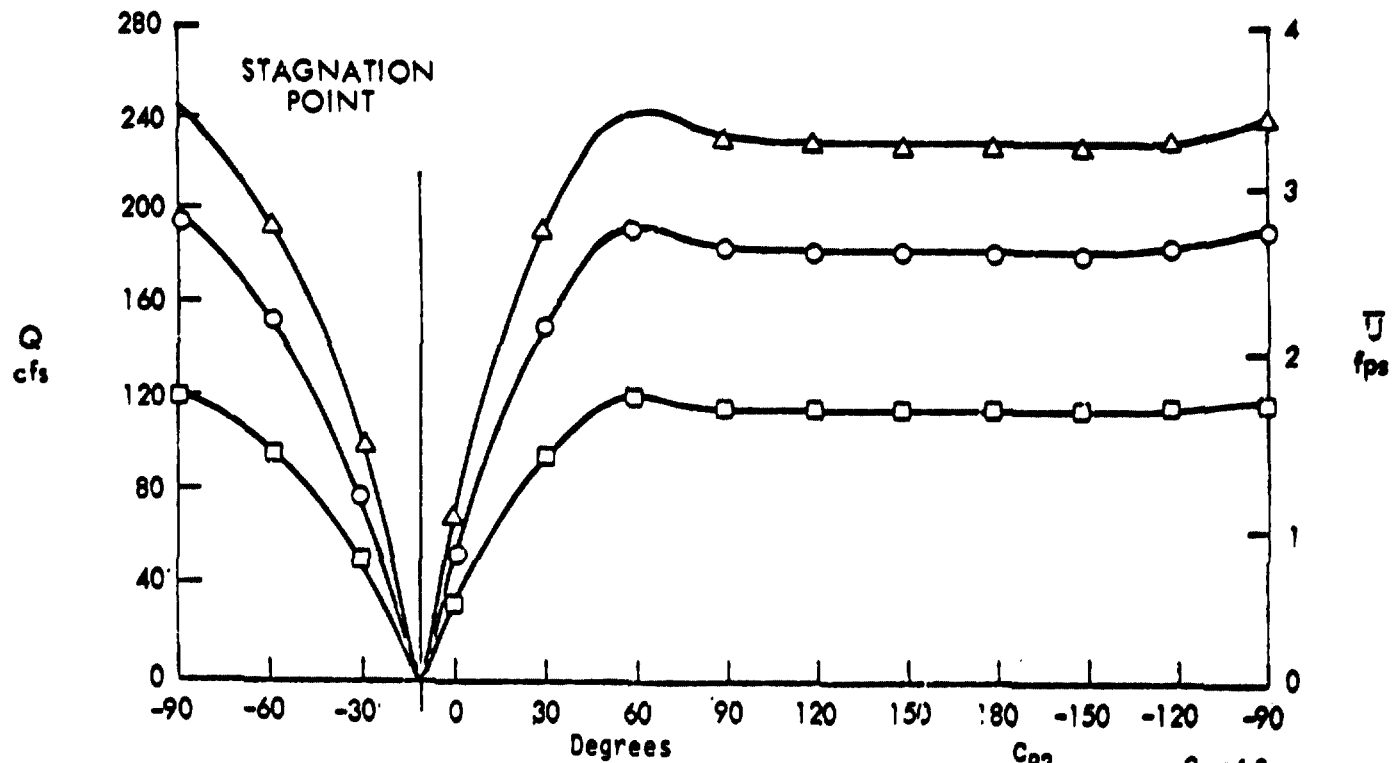


0.1' WAKE $C_{p1}(\theta) - C_{pw}$





VENTILATION PERFORMANCE IN 2-DIM. FLOW OVER FUSELAGE



$$C_1 = C_2 = 0.6$$

$$U_\infty = 10 \text{ mph}$$

$$A_1 = 2 \text{ m}^2$$

$$R_{D_0} = 4.8 \text{ E}4$$

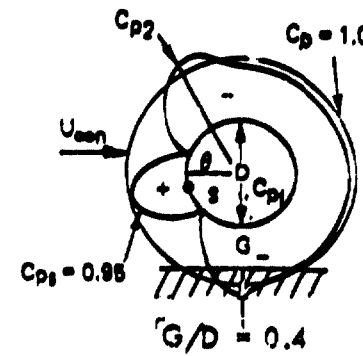
$$R_D = 1.23 \text{ E}6$$

$$A_2/A_1$$

$$\Delta 2.0$$

$$\circ 1.0$$

$$\square 0.5$$





RECOMMENDATIONS

- CONDUCT JSC FULL-SCALE FIRE TESTS TO VALIDATE THE ESTIMATES OF FUSELAGE VENTILATION OF THIS ANALYSIS
- FOR THE REAL WIND-ABOUT-FUSELAGE CONDITIONS EXPERIMENTALLY DETERMINE VENTILATION RATES APPLICABLE TO POST-CRASH FIRES & FULL-SCALE FIRE TESTS
 - WIND SPEED & DIRECTION
 - FULL-SCALE REYNOLDS NUMBERS
 - FUSELAGE SHAPE
 - FUSELAGE ORIENTATION & PROXIMITY TO GROUND
 - FUSELAGE-OPENINGS SIZE & LOCATION
 - FIRE-CONVECTION INDUCED SPEED OR CIRCULATION